Performance enhancement of Savonius wind turbine through partially deformable blades

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Abstract. In this study, we employ partially deformable blades to elevate the performance of Savonius wind turbines. The Bucket is constructed with rigid components equipped with a guidance system, resulting in continuous changes in its shape during turbine rotation. As the trailing edge of the advancing blade expands, it creates an active slot, effectively correcting the Bucket’s pressure distribution and enhancing the positive torque generated by the turbine. We employ a two-dimensional (2D) numerical model, implemented using the commercial software ANSYS-Fluent 23.0, with the governing motion equation executed through a user-defined function (UDF). This investigation explores the mechanism of performance enhancement by varying expansion amplitudes. Our results, obtained at a Tip-speed ratio (TSR) of 1, reveal that when the amplitude of deformation exceeds one-quarter of the Bucket radius, the partially deformable blade outperforms the rigid blade, leading to a remarkable 32% improvement in the torque coefficient. These findings signify a promising path toward enhancing Savonius turbine efficiency.

Keywords: Savonius turbine / deformable blade / wind turbine efficiency / harvesting wind energy

1 Introduction

Over the past decade, energy consumption has sharply increased, driven by population growth and technological advancements [1]. Meeting this rising energy demand with sustainable and limitless sources has become crucial for economic growth. To address this challenge, various systems for harnessing energy from renewable sources have been developed or enhanced [2]. Two natural resources that can be harnessed for energy are tides and wind, which can be conventionally captured using rotating turbines [3]. These turbines are categorized into horizontal turbines, with axes of rotation parallel to the incoming flow, and vertical turbines, with axes perpendicular to the flow. Among vertical turbines, there are Savonius turbines (drag-based) and Darrieus turbines (lift-based), along with some using combined lift and drag forces [4]. Numerous studies have aimed to improve the performance of Savonius turbines [5]. Most of these efforts have focused on modifying blade shapes, and thickness, or using baffles to alter their interaction with the incoming flow [6]. For example, placing a circular cylinder upstream of the returning blade can enhance performance by 12.2% at a specific tip speed ratio [7]. Additionally, using obstacles like curved circular plates in front of the blades has increased turbine performance by up to 19% [8]. Deformable blades have also been explored to reduce returning blade drag. Some researchers have proposed rigidly deformable blades that open when advancing and close when returning, resulting in a 25.9% improvement in performance [9]. Self-expandable blades that passively expand relative to the turbine angle have demonstrated the potential to increase the torque coefficient by 90.6% [10]. Studies involving elliptical and radii deformable blades have shown promise for novel energy extraction methods [11,12]. Vertical strips have been proposed to control the negative effects of high wind speed, effectively turning rigid blades into blades with numerous openings [13]. Hinged blades with a guiding system have improved efficiency by 52% [14]. Flexibility in turbine blades has also received attention. Tides with elastic blades that deform due to fluid loads have shown a remarkable 90% improvement compared to rigid turbines [15]. Tidal turbines with flexible blades, which open up in response to incident flow, have indicated the potential for performance enhancements under specific flow conditions [16].

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Table 1 compares the findings of this research with several other key studies in the area of flow control technology that have been previously published.

In summary, research and development efforts have explored various techniques, including blade modification, obstacle placement, and deformable blades, to enhance the performance of Savonius turbines and other vertical turbine designs, offering promising solutions to address energy challenges and promote sustainability.

This work suggests a new Savonius vertical axis wind turbine model with partially deformable blades. The present model has so far not been proposed by other authors. The effect of deformation amplitude and the length of expandable parts have been investigated. The results indicate that at a Tip-speed ratio (TSR) of 1, when the amplitude of deformation exceeds one-quarter of the Bucket radius, the performance of the partially deformable blade is significantly superior to that of the rigid blade. Specifically, the torque coefficient exhibits an impressive improvement of approximately 32%. The proposed investigation has the potential to significantly enhance the performance of Savonius wind turbines, opening the path to higher efficiency.

2 Description of the proposed model

Figure 1 depicts the suggested model, which is a Savonius vertical axis wind turbine with partially deformable blades. Each blade has two parts: a rigid part connected to the turbine axis, which performs a pure rotation motion, and a radii deformable part that has a combined rotation/deformation motion (see Figs. 2a and 2b). The Bucket form changes continuously during the turbine rotation. When the trailing edge of the advancing blade expands, an active slot is created. This slot achieves its maximum amplitude in every quarter of the cycle and is closed with a half-turbine rotation. The initial diameter of the turbine before deformation is 1 m (radius = 0.5 m). The swept area $A_s$ remains the same because the expansion and the contraction have the same amplitude. So, the power available for extraction can be expressed as:

$$P_w = \frac{1}{2} \rho A_s U_\infty^3.$$  (1)
The power coefficient can be defined as the ratio between the energy extracted by the turbine and the total obtainable energy [17–22]:

\[ CP = \frac{P_m}{P_w} = \frac{P_m}{\frac{1}{2} \rho A s U^3}. \tag{2} \]

The torque coefficient \( C_t \) is defined as:

\[ C_t = \frac{T}{\frac{1}{2} \rho A s U^2}. \tag{3} \]

In this examination, the frequency of expansion matches that of rotation, and the different amplitude of expansion ranged from \( A = 0.125 R_1 \) to \( A = 0.5 R_1 \) are examined, where \( R_1 \) represents the bucket radius. The rotor geometrical parameters are summarized in Table 2.

### Table 2. Turbine geometry parameters.

<table>
<thead>
<tr>
<th>Parameter</th>
<th>Value</th>
</tr>
</thead>
<tbody>
<tr>
<td>Rotor radius (R)</td>
<td>0.5 m</td>
</tr>
<tr>
<td>Height (H)</td>
<td>1 m</td>
</tr>
<tr>
<td>Area ( A_s )</td>
<td>1 m²</td>
</tr>
<tr>
<td>Blade thickness ( s )</td>
<td>0.0024 m</td>
</tr>
<tr>
<td>Bucket radius ( R_1 )</td>
<td>0.25 m</td>
</tr>
</tbody>
</table>

In this examination, the flow around rotating machinery involves flow separation, swirl effect, and adverse pressure gradient. Thus, to predict correctly such flow behavior, the turbulence model must be selected judiciously. The RNG k-\( \varepsilon \) model was used in this work. The RNG k-\( \varepsilon \) and SST k-w models are among the most frequently used models in CFD modeling of Savonius turbine [23–25]. The near-wall treatment was employed. The first grid cell requires being at about \( y+ \approx 1 \). A deformable computational mesh is displayed in Figure 3. The external domain dimensions are selected to ensure that flow is fully developed.

### 3.2 Motion equation

The 2D numerical simulation is approved using ANSYS FLUENT v18.1. The deformable turbine equation is executed within the DEFINE GRID MOTION macro. These equations are given as:

\[ x = x_{old} A \sin(\omega_1 t + \varphi), \tag{4} \]

\[ y = y_{old} A \sin(\omega_1 t + \varphi). \tag{5} \]

where \( y_{old} \) and \( x_{old} \) represent the blade position along \( x \) and \( y \)-axis in the previous time step. The angular velocity \( \omega_1 = 2\pi f_1 \) where \( f_1 \) is the deformation frequency.

### 3.3 Boundary conditions

The BC corresponds to the 2D computational domain presented in Figure 3. The time steps (\( \Delta t \)) are selected to be reliable with 0.5 degrees of rotation every one \( \Delta t \) [21–26]. The time steps are linked to the angular velocity \( \omega \) and accordingly to the tip speed ratio by equation (6).

\[ \Delta t = \frac{2\pi}{\omega} \cdot \frac{1}{360 \div 0.5}. \tag{6} \]

The used time steps are listed in Table 4.
4 Results

4.1 Grid independence test and model validation

The reliability of the mesh model is confirmed by testing three grids with different densities. Figure 4. Shows the results of the computational test. It can be seen that when the mesh size exceeds 100,000 cells, plots overlap with each other. Reasonably, the grid quantity test is proved. To reduce computation cost without sacrificing accuracy, the mesh size of 100,000 cells (420 nodes on the turbine blade) is used for all cases. The grid specifications of the present mesh are listed in Table 5:

Table 3. Boundary conditions.

<table>
<thead>
<tr>
<th>Boundary</th>
<th>Parameter</th>
<th>Value</th>
<th>Note</th>
</tr>
</thead>
<tbody>
<tr>
<td>Inlet</td>
<td>Constant velocity</td>
<td>7 m/s</td>
<td>Inlet turbulent viscosity = 0.001 m²/s</td>
</tr>
<tr>
<td>Outlet</td>
<td>Gauge pressure</td>
<td>0 Pa</td>
<td>Outlet turbulent viscosity = 0.001 m²/s</td>
</tr>
<tr>
<td>Sides</td>
<td>Symmetry</td>
<td>/</td>
<td>/</td>
</tr>
<tr>
<td>Blades</td>
<td>Wall</td>
<td>/</td>
<td>No slip wall condition</td>
</tr>
</tbody>
</table>

Table 4. Time steps versus tip speed ratio.

<table>
<thead>
<tr>
<th>TSR[−]</th>
<th>ω[rad/s]</th>
<th>Δ t[s]</th>
</tr>
</thead>
<tbody>
<tr>
<td>0.4</td>
<td>5.6</td>
<td>1.5.10^{-3}</td>
</tr>
<tr>
<td>0.7</td>
<td>9.8</td>
<td>8.9.10^{-4}</td>
</tr>
<tr>
<td>1</td>
<td>14</td>
<td>6.2.10^{-4}</td>
</tr>
</tbody>
</table>

4.2 Comparative analysis between deformable and conventional turbine

Figure 6 depicts the individual torque coefficient for rigid and deformable blade case $a = 0.375 \ R_1$, TSR = 1. Todescribe the flow behavior sufficiently during the turbine rotation, pressure contour for different instants is provided. Results reveal that the hinged or deformable turbine tends to reduce the torque fluctuation thus; the torque peaks of the conventional turbine are greater than that of the deformable blade this is attributed to the fact that at the onset of the deformation, a slot (void) is formed between...
Table 5. Grid specifications of the present mesh.

<table>
<thead>
<tr>
<th>Mesh density</th>
<th>Coarse</th>
<th>Medium</th>
<th>Fine</th>
</tr>
</thead>
<tbody>
<tr>
<td>Total number of cells</td>
<td>80,000</td>
<td>100,000</td>
<td>120,000</td>
</tr>
<tr>
<td>Number of cells in the near wake zone</td>
<td>50,000</td>
<td>60,000</td>
<td>70,000</td>
</tr>
<tr>
<td>Number of cells in the far wake zone</td>
<td>30,000</td>
<td>40,000</td>
<td>50,000</td>
</tr>
<tr>
<td>First layer thickness (mm)</td>
<td>0.5</td>
<td>0.1</td>
<td>0.05</td>
</tr>
<tr>
<td>Number of nodes on the turbine blade</td>
<td>260</td>
<td>420</td>
<td>630</td>
</tr>
<tr>
<td>Average y+</td>
<td>2.1</td>
<td>0.4</td>
<td>0.3</td>
</tr>
</tbody>
</table>

Fig. 5. CFD model compared to experiment data [27] (a) torque coefficient (b) power coefficient.

Fig. 6. Comparison of the individual torque coefficient for rigid and deformable blade case $A = 0.375 R_1$ and Pressure contour for different instants for rigid and expandable blade.
Fig. 7. Effect of expansion amplitude on the instantaneous torque coefficient of (a) blade A (b) blade B and (c) the total torque coefficient at TSR = 1.

Fig. 8. Effect of expansion amplitude on (a) Average torque coefficient CT (b) Average Power coefficient CP.
the blade rigid and deformable parts, which reduces the torque peak \((\theta < 75^\circ)\). A supplementary suction zone is formed near the slot of the advancing blade 1, which shifts the appearance of negative torque from \(\theta = 105^\circ\) in the nominal case to \(\theta = 205^\circ\) for the deformable blade. For the hinged (deformable) blade 2, the contraction of the deformable part generates a smaller arm, which advantageously reduces the returning blade drag.

4.3 Effect of expansion amplitude on the turbine performance

The effects of deformation amplitude on the individual and total instantaneous torque coefficient are shown in Figure 7. For the advancing blade 1, it can be seen that the amplitude of 0.5 \(R_1\) displays the widest \(C_m\) region, which covers the azimuthal angle between \(80^\circ < \theta < 300^\circ\). By reducing the

![Figure 9. Effect of deformable part length on the instantaneous torque coefficient of (a) blade A (b) blade B and (c) the total torque coefficient for \(A = 0.375\ R_1\) at TSR = 1.](image-url)
amplitude from 0.5 $R_1$ to 0.125 $R_1$, the $C_m$ of blade1 converges to that of the nominal blade. For the returning blade 2, it can be seen that the deformation is beneficial in the first third of the cycle ($\theta < 120^\circ$). For $120^\circ < \theta < 250^\circ$, the returning rigid blade performs better than all deformable blades. This is attributed to the energy loss due to the blade expansion. For $\theta > 250^\circ$ higher torque is provided by all deformable blades due to the larger blade arm. From Figures 7c and 7d, the maximum $C_m$ of 0.341 is reached for a blade with the amplitude of 0.5 $R_1$, which is about 32% times that of a rigid blade.

By looking at the time-mean torque coefficient shown in Figures 7c and 7d, a decrease of 7.85% $C_m$, as compared to the rigid blade, is noted with an amplitude of 0.125 $R_1$. As stated previously, the flow control mechanism is associated with the extension and/or the increase in strength of the suction zone along the advancing blade. The appearance of slot breaks down this suction zone, which affect negatively the performance at low deformation amplitudes.

Figures 8a and 8b displays the effect of deformation amplitude on the average torque and Power coefficients. It can be seen that at TSR = 0.4 the flow control using the present technique is not beneficial. At low TSR, The appearance of slot breakdowns this suction zone, which affect negatively the performance at low deformation amplitudes. Figures 9 shows the effect of deformable part length on the instantaneous torque coefficient for $A = 0.375 R_1$ at TSR = 1. From the previous discussion on how the appearance of slot breakdown the suction zone, it was expected that, the blade with a smaller arc length $L = 1/4 R_2$ provide higher $C_m$. Because in this case, the turbine will benefit from a longer arm without disturbing the suction zone. The impact of slot location on the flow structure can be also obtained from the pressure contour plots in Figures 10. It can be noted that another suction zone is formed near the slot of the advancing blade for both cases. However, in the case $L = 1/4 R_2$, the suction zone is more intense and does not affect the trailing edge vortex. This behavior increases the gain of improvement from 13.2% in the case of $L = 1/2 R_2$ to 18.9% in the case of $L = 1/4 R_2$.

5 Conclusion

In this study, a Two-dimensional (2D) numerical model of a deformable Savonius turbine is approved. Partially deformable blades are used to enhance the turbine performance. The Bucket has a time-varying form that changes continuously during the turbine rotation. The trailing edge of the blades expands or contracts relative to the turbine’s azimuthal position. An active slot is created, which affects the blade pressure distribution.

The main conclusions can be listed as:
- The flow control mechanism is associated with the extension and/or the increase in strength of the suction zone along the advancing blade.
- The blade with a smaller expandable arc length $L = 1/4 R_2$ provides higher $C_m$.
- The maximum $C_m$ of 0.341 is reached for a blade with an amplitude of 0.5 $R_1$, which is about 32% times that of a rigid blade.
- The time-mean torque coefficient shows, a decrease of 7.85% $C_m$, as compared to rigid blade, for an amplitude of 0.125 $R_1$.

4.4 Effect of deformable part length

This section studied the effect of two arc lengths $L = 1/4 R_2$ and $L = 1/2 R_2$ where $R_2$ is the bucket circular arc length. Figures 9 shows the effect of deformable part length on the instantaneous torque coefficient. From the previous discussion on how the appearance of slot breakdown the suction zone, it was expected that, the blade with a smaller arc length $L = 1/4 R_2$ provide higher $C_m$. Because in this case, the turbine will benefit from a longer arm without disturbing the suction zone. The impact of slot location on the flow structure can be also obtained from the pressure contour plots in Figures 10. It can be noted that another suction zone is formed near the slot of the advancing blade for both cases. However, in the case $L = 1/4 R_2$, the suction zone is more intense and does not affect the trailing edge vortex. This behavior increases the gain of improvement from 13.2% in the case of $L = 1/2 R_2$ to 18.9% in the case of $L = 1/4 R_2$. 

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- The time-mean torque coefficient shows, a decrease of 7.85% $C_m$, as compared to rigid blade, for an amplitude of 0.125 $R_1$. 

Fig. 10. Comparison of pressure contours.
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